

# Stennis Space Center (SSC) B-1 Test Facility Liquid Hydrogen Run Tank Vacuum Anomaly & Repair

Nicholas J. Nugent April 10, 2012

Engineering & Test Directorate NASA, Stennis Space Center



### LH Tank Background

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- 90,000 gallon, 66 psig
- Built in 1962 by Chicago Bridge and Iron Works Company for SSFL
- Relocated from SSFL to SSC in 1984 by barge
- Inner vessel shifted in 1985 during vessel modifications and repaired before being put into service at SSC
- New 12" Run and 8" Fill and Drain Nozzles added and existing nozzles removed in 1985
- Put into service on the B-1 Test Facility in 1988 in support of SSME testing
- Last recertified in June 2004 and is due for recertification in 2014









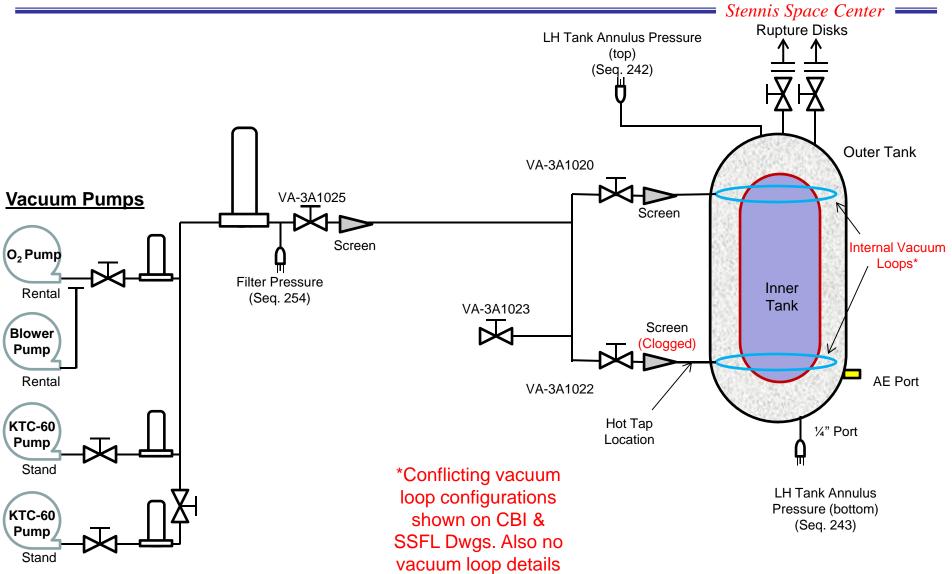
1985 Vessel on barge at SSC



1985 Damage to Manway Liner

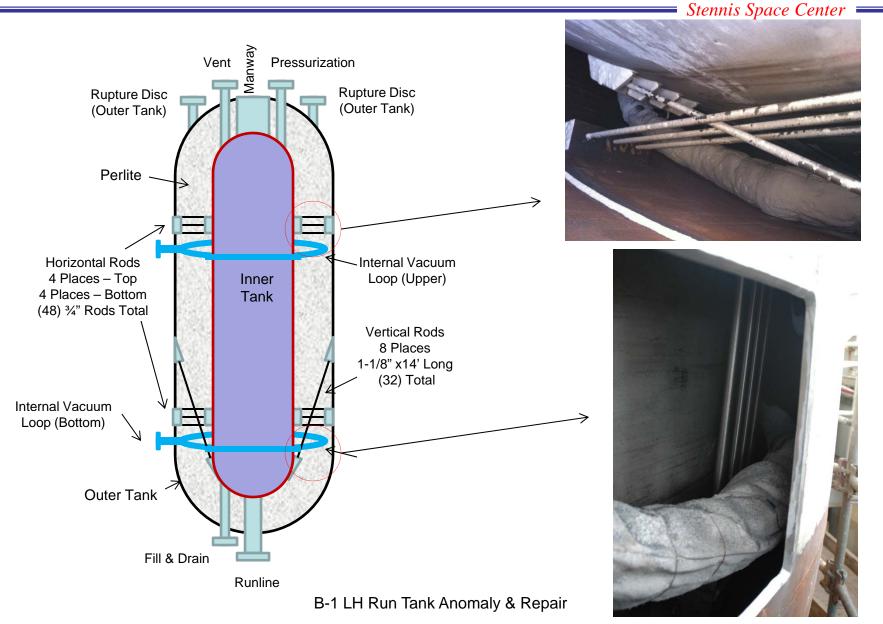


## B-1 LH Tank Vacuum System Configuration





### **B-1** LH Tank Inner Details



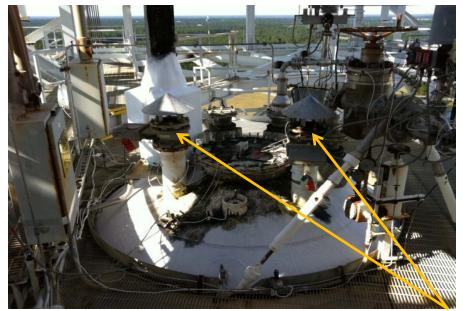


## Anomaly Timeline

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#### •October 21, 2011

- Vacuum issue found
  - Top of the tank high (>20,000 microns)
  - Bottom of the tank nominal (<10 microns)</li>
  - Liquid level ~50%
- Cracks discovered in the two rupture disc holders located on top of the vessel



LH Run Tank (top)

Rupture Discs & Holders

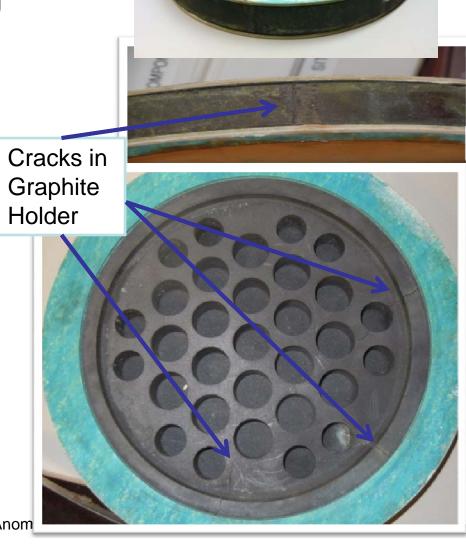
#### Scenario:

- The cracks allowed an unknown amount of air to enter the annulus.
- The air then froze to the inner vessel outer wall.
- The differential vacuum pressure between the top and bottom annulus indicated compacted perlite and/or frozen air in the annulus.
- High top pressure cause still under investigation. Likely the result of crack growth since previous measurement and/or a vacuum pump being switched off/serviced preceding weekly measurement.



## Rupture Discs

- Rupture disc #1 was identified to be leaking via bagging since cracks not visible when installed
- Rupture disc was removed and found to be cracked
- Rupture disc holder was made of graphite and installed in 2003.
- Companion flange was heavily corroded
  - Wire brushing of flange was unsuccessful
  - Flange was sent to machine shop and faced for improved finishing and sealing
- Companion flange and new rupture disc installed
- 2<sup>nd</sup> rupture disc also leaking but could not be removed due to leaking isolation valve. Utilized Kenny Seal to stop leak.
- No leakage detected via bagging method





## Anomaly Timeline (Preps for Hotfire)

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- Customer required one more engine hotfire test prior to a three month scheduled outage
- Attempted to remove air (nitrogen and oxygen) in annulus by vaporizing frozen air in annulus:
  - Rented large blower vacuum pump
  - Pressurized inner tank to warm-up tank wall
    - •Frozen air sublimated as vessel wall temp rose
  - Monitored/controlled top annulus pressure
- Discovered elevated O2 in vacuum pump discharge. Rented O2 rated pump.
  - •After ~24 hrs top annulus pressure dropped
- Conducted tank pressurizations tests to show minimal top annulus vacuum rise (< 5 psia)</li>
- Filled run tank to 100%...top annulus dropped from 1.4 psia (~70,000 microns) to 23 microns
- · Set observer cutoff limits on top annulus readings
- Successfully completed two engine hotfires
  - November 2<sup>nd</sup>
  - November 10<sup>th</sup>

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Rental Pumps



## B-1 & A-2 LH Tank VJ Test Data Comparison

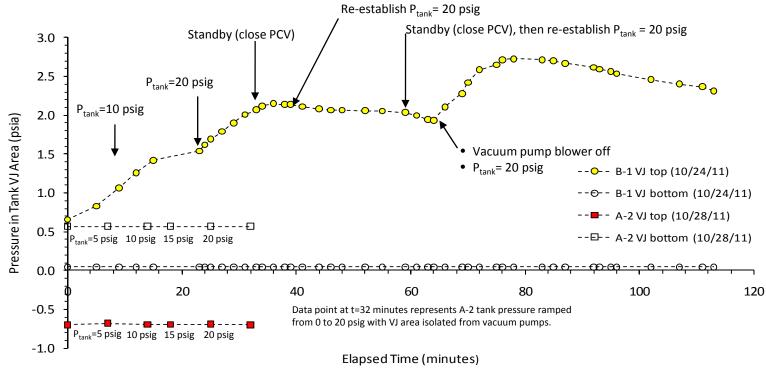
Stennis Space Center =

#### **Test Objective**

The B-1 and A-2 LH run tanks were pressurized in a series of controlled steps (5 psig, 10 psig, 15 psig & 20 psig)
 while monitoring the pressures in the vacuum jacket (VJ) area

#### <u>Results</u>

- The pressures in the A-2 LH tank VJ area did not change as the LH run tank was pressurized from 0 to 20 psig
- The A-2 results significantly differ from B-1 adding evidence of a air filled annulus
- Care and precautions must be taken during detanking due to the frozen annulus



Test Notes: Elapsed Time (minutes)

• A-2 test completed on 10/28/11 using TPS A2-00891; A-2 LH Tank Level ~ 56% full & B-1 LH Tank Level ~ 44% full.

Vacuum pump exhaust before and after test contained ~ 10.2% O<sub>2</sub>; 0% LEL.



## Anomaly Timeline (Detanking)

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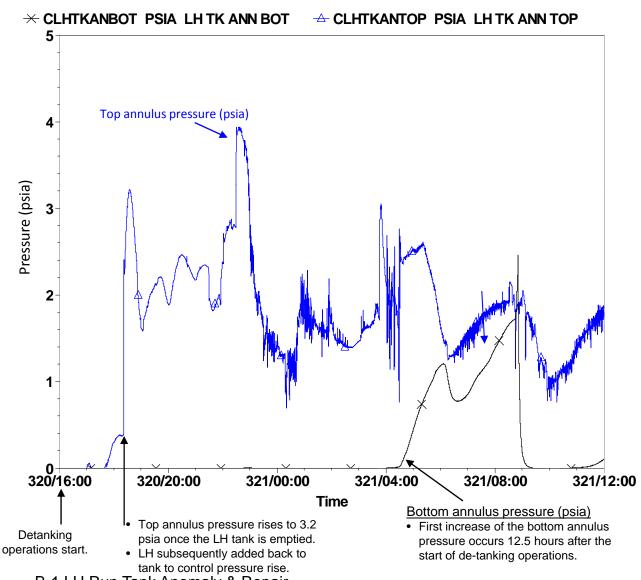
- Precautions were established to drain the tank after the last engine test :
  - Vacuum differential pressure limit established between the top and bottom of the annulus to prevent shifting/damage of the inner vessel (2.5 psid bottom over top limit)
  - Slow emptying of the vessel with continuous vacuum pumping on the top and bottom annulus
  - Two LH barges were stationed at the test facility to refill the LH vessel to lower the vacuum pressure increase if needed
- Began detanking on November 16, 2011
- A combined customer, NASA/SSC, and Test Operations Contract (TOC) team conducted the operations on a 24 hour/7 day a week basis
- Top annulus pressure began to rise.
- 12.5 hours after detanking bottom annulus pressures began to rise
- Bottom vacuum header proved ineffective in controlling the bottom annulus pressure indicating the bottom vacuum header was non functional. Top header had no effect on bottom annulus.
- Contingency processes were implemented to transfer small amounts of liquid hydrogen back into the vessel as needed to lower vessel annulus pressures



## Annulus Pressures During Detanking

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- B-1 LH detanking operations commenced on 11/16/11 (day 320) at 4:00 pm.
- Generally, pressure-vent cycles were performed in addition to emptying the inner tank of I H.
- During the first day of detanking, the top annulus pressure rose sharply to 3.2 psia once the LH tank was empty while the bottom annulus pressure remained at 0 psia.
- The first increase of the bottom annulus pressure occurred on 11/17/11 at 4:30 am, 12.5 hours after the start of de-tanking operations.
- Pressure-vent cycle at 10:45 am (11/17/11) yielded a vent gas temperature of ~140°R.





## Anomaly Timeline (Frost/Ice Buildup)

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#### **November 18, 2011**

- Two days after detanking heavy ice/frost developed on the bottom of the outer vessel
- The vessel bottom temperature continued to drop into the 140-160 Rankin, consistent with liquid air temperatures
- Well below the 410 Rankin ductility range for the carbon steel material subjecting the outer vessel to cracking at any time



Bottom of LH Run Tank at Low Temperature



## Anomaly Timeline (Attempts to Establish Pumping Capability)

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- Attempted to establish pumping ability on bottom annulus
- Utilized existing instrumentation outlet on the bottom of the outer vessel
- Resulted in limited pumping capability due to the small (1/4") available connection size
- AE port
- Upon initiation of pumping vacuum, large amounts of perlite were pumped from the annulus resulting in continuous cleaning of facility vacuum system filters
- "Hot tap" drilling device was designed, fabricated on site, and utilized to drill into the existing bottom vacuum header immediately adjacent to the outer vessel bypassing any clogged external filters



1/4" Instrumentation Port used for Vacuum Service



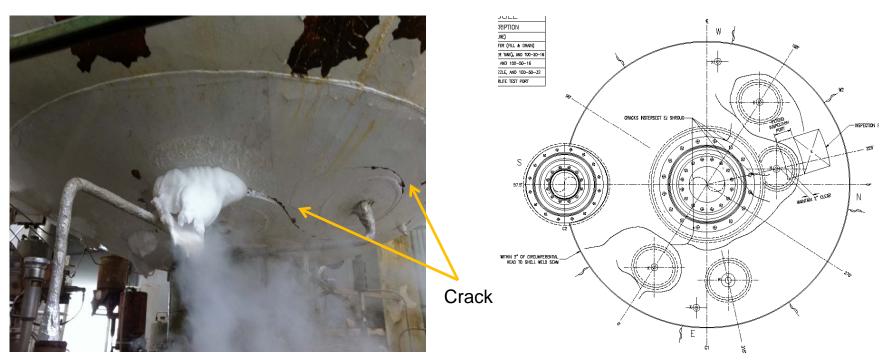
Plug Removed for Vacuum Connection to AE port



## Anomaly Timeline (Rupture of Bottom Head)

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- At 10:05am on December 4, 2011, during preparation for installation of the external perlite filtering system, an audible "pop" emanating from below the tank was heard by the test crew
- The on stand test crew was evacuated into the protective concrete hard core of the test facility and then evacuated to the test control center through the instrumentation tunnel
- Video and instrumentation data at the test control center showed a release of vapor from the bottom of the vessel through a crack accompanied by an immediate slow rise in the vacuum annulus pressure at the bottom of the vessel.

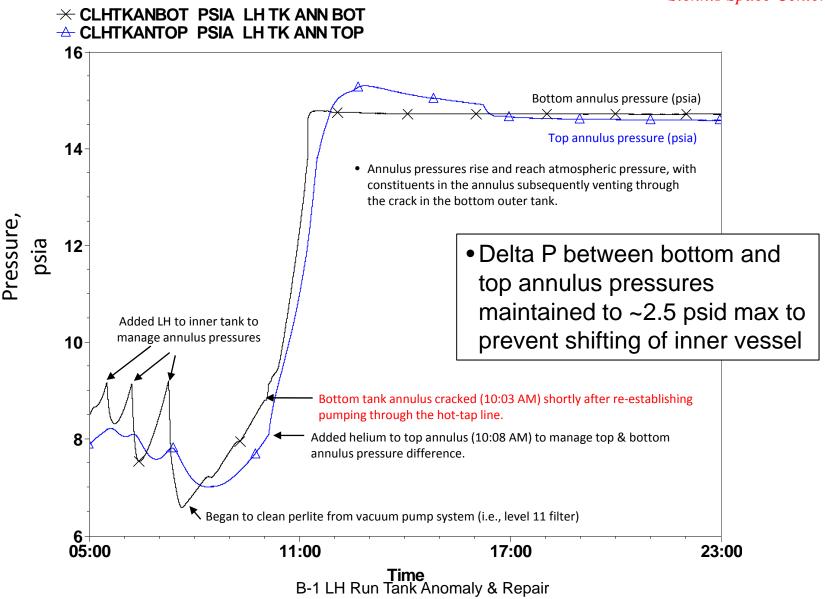


**LH Tank Bottom** 



### Crack in Outer Tank Timeline

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### What We've Learned

#### Stennis Space Center ——

- Lower Vacuum Header
  - Physically disconnected inside annulus
  - Made from flexible hose
  - Lower witch's hat screen plugged
- Previous Issues with Tank at SSFL (prior to 1970)
  - Vacuum issues: 6000 micron minimum
  - Iced over vessel bottom during detanking
  - Cracked bottom head area
- Compacted Perlite
- Full vessel will cryopump annulus and show good vacuum
- Inner Vessel
  - Successful Acoustic Emissions test
  - Successful structural support inspections
- Outer Vessel

Numerous cracks identified in smaller lower head No issues found in main lower head







**Lower Vacuum Header Disconnected** 



**Vacuum Header Flexible Hose** 

Witch's Hat Screen



## B-1 LH Tank Vacuum System Internal Manifold Redesign

#### **Concept**

- 4 internal vertical manifolds
- Suspended on spring on one end, welded on other end
- Sections of perforated or slotted tubing covered with sock filtration material

#### **Pros**

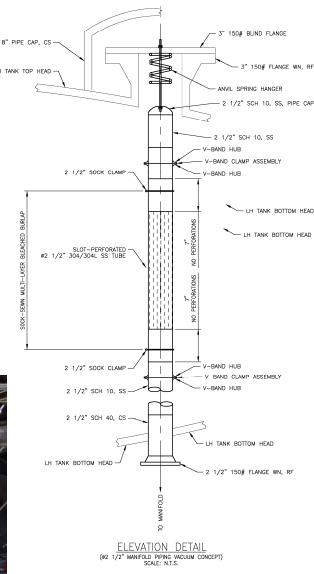
- High surface area (greater than current B-1 system)
- Manifold traverses tank vertically allowing internal equalization of pressure on top and bottom
- Significantly fewer tank penetrations (less expensive, faster, and few welds that could leak)

#### **Cons**

- No experience on site with vertical vacuum manifold
  - Industry contacts confirm use of vertical vacuum manifolds is not uncommon



#### Stennis Space Center •





## Future Improvements

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- Rupture disc replaced by lift plates
- Preventative maintenance on lift plates
- Weld or install weld caps on mechanical joints where possible (i.e., perlite fill nozzles)
- New vacuum header design
  - Allow better top/bottom communication (i.e pressure can equalize in the annulus)
  - Better accessibility to components
- Prior to future detankings
  - Verify no massive air in leakage by monitoring top annulus pressure while performing pressurization cycles
- Better vacuum monitoring
  - Remote top and bottom readings
  - Automated alarms
- Standardize pumping operations
  - Isolate vacuum system post LH loading
  - If pumps are needed then investigate for air in leakage



## Questions?